Issued in Kansas City, Missouri, on May 25, 1995.

Henry A. Armstrong,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service. [FR Doc. 95–13468 Filed 6–1–95; 8:45 am] BILLING CODE 4910–13–U

14 CFR Part 39

[Docket No. 94-NM-240-AD; Amendment 39-9255; AD 95-12-05]

Airworthiness Directives; Lockheed Model 382 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.
ACTION: Final rule.

SUMMARY: This amendment supersedes an existing airworthiness directive (AD), applicable to certain Lockheed Model 382 series airplanes, that currently requires a revision to the Airplane Flight Manual to require takeoff operation in accordance with revised performance data. This amendment requires installation of certain valve housings for the propeller governor on the outboard engines. This amendment is prompted by a report of a change that had been incorporated into the propeller governor of these airplanes during production, which altered the thrust decay characteristic of the propeller when operating in an engine failure scenario. The actions specified by this AD are intended to ensure that the airplane maintains adequate thrust decay characteristics in the event of critical engine failure during takeoff. DATES: Effective July 3, 1995.

The incorporation by reference of Lockheed Airplane Flight Manual Supplement 382–16, dated August 11, 1993, as listed in the regulations, was approved previously by the Director of the Federal Register as of August 10, 1994 (59 FR 35236, July 11, 1994).

ADDRESSES: The service information referenced in this AD may be obtained from Lockheed Aeronautical Systems Support Company (LASSC), Field Support Department, Dept. 693, Zone 0755, 2251 Lake Park Drive, Smyrna, Georgia 30080. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue SW., Renton, Washington; or at the FAA, Atlanta Aircraft Certification Office, Small Airplane Directorate, Campus Building, 1701 Columbia Avenue, Suite 2-160, College Park, Georgia; or at the Office of the Federal Register, 800 North Capitol Street NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Thomas Peters, Aerospace Engineer, FAA, Flight Test Branch, ACE–160, Small Airplane Directorate, Atlanta Aircraft Certification Office, Campus Building, 1701 Columbia Avenue, Suite 2–160, College Park, Georgia 30337–2748; telephone (404) 305–7367; fax (404) 305–7348.

SUPPLEMENTARY INFORMATION: A

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 94–14–09, amendment 39–8961 (59 FR 35236, July 11, 1994), which is applicable to certain Lockheed Model 382 series airplanes, was published in the **Federal Register** on February 8, 1995 (60 FR 7480). The action proposed to require removal of any servo-type valve housing assembly, having part number 714325–2, –3, –5, –6, or –7, installed on any outboard engine, and replacement of those assemblies with part number 714325–1.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the single comment received.

The commenter supports the proposed rule.

After careful review of the available data, including the comment noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

There are approximately 112 Model 382, 382E, and 382G series airplanes of the affected design in the worldwide fleet. The FAA estimates that 18 airplanes of U.S. registry will be affected by this AD, that it will take approximately 8 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will cost approximately \$90,000 per airplane. Based on these figures, the total cost impact of the AD on U.S. operators is estimated to be \$1,628,640, or \$90,480 per airplane.

The FAA has been advised that the only U.S. operator of Lockheed Model 382 series airplanes has already equipped half of its fleet (9 airplanes) with the valve housing assembly that will be required by this rule. Therefore, the future economic cost of this rule on U.S. operators is now only \$814,320.

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does

not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a ''significant rule'' under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. App. 1354(a), 1421 and 1423; 49 U.S.C. 106(g); and 14 CFR 11.89.

§ 39.13 [Amended]

2. Section 39.13 is amended by removing amendment 39–8961 (59 FR 35236, July 11, 1994), and by adding a new airworthiness directive (AD), amendment 39–9255, to read as follows:

95–12–05 Lockheed: Amendment 39–9255. Docket 94–NM–240–AD. Supersedes AD 94–14–09, Amendment 39–8961.

Applicability: Model 382, 382E, and 382G series airplanes; equipped with a servo-type valve housing assembly, having part number 714325–2, –3, –5, –6, or –7, installed on any outboard engine; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must use the authority provided in paragraph (c) to request approval from the FAA. This approval may address either no action, if the current configuration eliminates the unsafe condition; or different actions necessary to address the unsafe

condition described in this AD. Such a request should include an assessment of the effect of the changed configuration on the unsafe condition addressed by this AD. In no case does the presence of any modification, alteration, or repair remove any airplane from the applicability of this AD.

Compliance: Required as indicated, unless accomplished previously.

To ensure that the airplane maintains adequate thrust decay characteristics in the event of critical engine failure during takeoff, accomplish the following:

- (a) Within 60 days after August 10, 1994 (the effective date of AD 94–14–09, amendment 39–8961), revise the Limitations and Performance Data Sections of the FAA-approved Airplane Flight Manual (AFM) to include information specified in Lockheed Airplane Flight Manual Supplement 382–16, dated August 11, 1993, and operate the airplane accordingly thereafter. The requirements of this paragraph may be accomplished by inserting AFM Supplement 382–16 into the AFM.
- (b) Within 24 months after the effective date of this AD, replace the servo-type valve housing assemblies having part number 714325–2, -3, -5, -6, or -7, with part number 714325–1, on the propeller governors installed on the outboard engines, in accordance with Lockheed Document SMP–515C, Card No. CO–135. Replacement of these assemblies with part number 714325–1, constitutes terminating action for the requirements of paragraph (a) of this AD; once the replacement is accomplished, the AFM revision may be removed.

Note 2: Propeller governors with servo-type valve housing assemblies having part number 714325–2, -3, -5, -6, or -7, may be retained or replaced with part number 714325–1 for use on the inboard engine positions.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Atlanta Aircraft Certification Office (ACO), FAA, Small Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Atlanta ACO.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Atlanta ACO.

- (d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (e) The AFM revision shall be done in accordance with Lockheed Airplane Flight Manual Supplement 382–16, dated August 11, 1993. The incorporation by reference of this document was approved previously by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51 as of August 10, 1994 (59 FR 35236, July 11, 1994). Copies may be obtained from Lockheed Aeronautical Systems Support Company (LASSC), Field Support Department, Dept. 693, Zone 0755, 2251 Lake

Park Drive, Smyrna, Georgia 30080. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Atlanta Aircraft Certification Office, Small Airplane Directorate, Campus Building, 1701 Columbia Avenue, Suite 2–160, College Park, Georgia; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(f) This amendment becomes effective on July 3, 1995.

Issued in Renton, Washington, on May 26, 1995

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 95–13505 Filed 6–1–95; 8:45 am] BILLING CODE 4910–13–U

14 CFR Part 71

[Airspace Docket No. 95-ACE-6]

Alteration of Class E Airspace Area; St. Louis, MO

AGENCY: Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule; correction.

SUMMARY: This document corrects a final rule published on May 3, 1995, that inadvertently removed the St. Louis, MO, Class E5 airspace designation. This action reflects the FAA's original intent to revise the St. Louis, MO, Class E5 airspace designation to exclude the Weiss Municipal Airport from the airspace designation. This action is a result of the closure of the Weiss Municipal Airport. EFFECTIVE DATE: 0901 UTC, May 3, 1995.

FOR FURTHER INFORMATION CONTACT:

Patricia P. Crawford, Airspace and Obstruction Evaluation Branch (ATP–240), Airspace-Rules and Aeronautical Information Division, Air Traffic Rules and Procedures Service, Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone: (202) 267–9255.

SUPPLEMENTARY INFORMATION: On May 3, 1995, the FAA published a final rule that removed the St. Louis, MO, Class E5 airspace designation (60 FR 21700). However, that action inadvertently removed the St. Louis, MO, Class E5 airspace area. This action reflects the FAA's original intent to revise the St. Louis, MO, Class E5 airspace designation to exclude the Weiss Municipal Airport from the airspace designation.

Correction of Final Rule

Accordingly, pursuant to the authority delegated to me, the

publication in the **Federal Register** on May 3, 1995 (60 FR 21700, **Federal Register** Document 95–10772), and the corresponding description in FAA Order 7400.9B, which is incorporated by reference in 14 CFR 71.1, are corrected as follows:

§71.1 [Corrected]

Charles, MO

*

ACE MO E5 St. Louis, MO [Revised] Lambert-St. Louis International Airport (Lat. 38°44′51″ N, long. 90°21′36″ W) Spirit of St. Louis Airport, MO

(Lat. 38°39′43″ N, long. 90°39′00″ W) St. Louis Regional Airport, Alton, IL (Lat. 38°53′25″ N, long. 90°02′45″ W) St. Charles County Smartt Airport, St.

(Lat. $38^{\circ}55'47''$ N, long. $90^{\circ}25'47''$ W) St. Louis VORTAC

(Lat. $38^{\circ}51'38''$ N, long. $90^{\circ}28'57''$ W) Foristell VORTAC

(Lat. 38°41′40″ N, long. 90°58′17″ W) ZUMAY LOM

(Lat. 38°47′17" N, long. 90°16′44" W) OBLIO LOM

(Lat. 38°48′01″ N, long. 90°28′29″ W) Civic Memorial NDB

(Lat. 38°53'32" N, long. 90°03'23" W)

That airspace extending upward from 700 feet above the surface within a 6-mile radius of the Lambert-St. Louis International Airport and within 4 miles southeast and 7 miles northwest of the Lambert-St. Louis International Airport Runway 24 ILS localizer course extending from the airport to 10.5 miles northeast of the ZUMAY LOM and within 4 miles southwest and 7.9 miles northeast of the Lambert-St. Louis Airport Runway 12R ILS localizer course extending from the airport to 10.5 miles northwest of the OBLIO LOM and within 4 miles southwest and 7.9 miles northeast of the Lambert-St. Louis Airport Runway 30L ILS localizer southeast course extending from the airport to 8.7 miles southeast of the airport and within a 6-mile radius of Spirit of St. Louis Airport and within 2.6 miles each side of the 098° radial of the Foristell VORTAC extending from the 6-mile radius area to 8.3 miles west of the airport and within a 6-mile radius of St. Charles County Smartt Airport, and within a 6-mile radius of St. Louis Regional Airport, and within 4 miles each side of the $0\bar{1}4^\circ$ bearing from the Civic Memorial NDB extending from the 6-mile radius to 7 miles north of the airport and within 4.4 miles each side of the 190° radial of the St. Louis VORTAC extending from 2 miles south of the VORTAC to 22.1 miles south of the VORTAC.

Issued in Washington, DC, on May 26, 1995

Harold W. Becker,

Manager, Airspace—Rules and Aeronautical Information Division.

[FR Doc. 95–13456 Filed 5–26–95; 3:59 pm] BILLING CODE 4910–13–P